

Lecture 4 on 9.4.2024: Subsonic ^{or supersonic} compressible flow around a thin airfoil at zero lift

A) Governing equations for compressible potential flow:

• Momentum + Continuity:

$$(C^2 - \phi_x^2) \phi_{xx} + (C^2 - \phi_y^2) \phi_{yy} - 2\phi_x \phi_y \phi_{xy} = 0 \quad (1)$$

• Total energy + thermodynamic relationships:

$$C^2 = C_\infty^2 + \frac{\kappa - 1}{2} (U^2 - \phi_x^2 - \phi_y^2) \quad (2)$$

where

$$\nabla \phi = \underline{u},$$

$$\kappa = \frac{c_p}{c_v}$$

Non-linear eq. $\rightarrow$ can be solved numerically (FEM / FD)

B) Dimensionless form: - for simplification & generality

$\rightarrow$ define dimensionless variables:

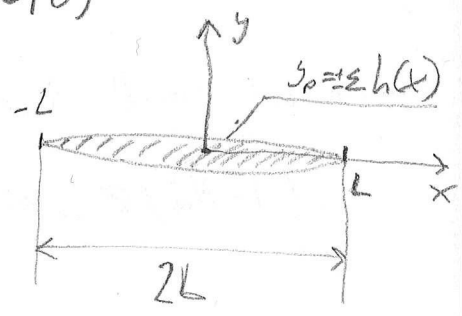
$$\tilde{x} = \frac{x}{L}, \quad \tilde{u} = \frac{u}{U}, \quad \tilde{C} = \frac{C}{C_\infty}; \quad \tilde{\phi} = \frac{\phi}{UL} \quad \text{s.t.} \quad \tilde{u} = \nabla \tilde{\phi} \quad (3)$$

$\rightarrow$ dimensionless eq.:

$$\left[M_\infty^{-2} + \frac{\kappa - 1}{2} (1 - \tilde{\phi}_y^2) - \frac{\kappa + 1}{2} \tilde{\phi}_x^2 \right] \tilde{\phi}_{\tilde{x}\tilde{x}} + \left[M_\infty^{-2} + \frac{\kappa - 1}{2} (1 - \tilde{\phi}_x^2) - \frac{\kappa + 1}{2} \tilde{\phi}_y^2 \right] \tilde{\phi}_{\tilde{y}\tilde{y}} - 2\tilde{\phi}_x \tilde{\phi}_y \tilde{\phi}_{\tilde{x}\tilde{y}} = 0 \quad (4)$$

c) Recall the problem formulation:

$\underline{u}_\infty = (U, 0)$



• Boundary conditions:

$\underline{\tilde{u}} \rightarrow (1, 0) \text{ as } \tilde{x}^2 + \tilde{y}^2 \rightarrow \infty$
 $\Rightarrow \tilde{\phi} = \tilde{\chi} \text{ as } \tilde{x}^2 + \tilde{y}^2 \rightarrow \infty \quad (5)$
 "far-field"

- symmetry: $\tilde{\phi}_{\tilde{y}} = 0 \text{ at } \tilde{y} = 0 \text{ for } \tilde{x} \leq -1 \text{ or } \tilde{x} \geq 1$
 - kinematic condition:

$\boxed{\frac{\tilde{\phi}}{\tilde{u}} = \frac{\tilde{\phi}_{\tilde{y}}}{\tilde{\phi}_{\tilde{x}}} = \pm \varepsilon \frac{\partial h}{\partial \tilde{x}}}$ at $\tilde{y} = \pm \varepsilon \tilde{h}$ for $-1 < \tilde{x} < 1$

d) Asymptotic expansion: (linearization)

- Assumption: the thin profile causes a small perturbation of the incoming flow, proportional to $\varepsilon \ll 1$

$\tilde{\phi} = \phi_0 + \varepsilon \phi_1$ (drop the $\sim$ for brevity) (8)

$\phi_0 = \tilde{\chi}$ ← undisturbed flow $\underline{u}_0 = (1, 0)$

$\tilde{\phi} = \tilde{\chi} + \varepsilon \phi_1$
 $\tilde{\phi}_{\tilde{x}} = 1 + \varepsilon \phi_{1,\tilde{x}}$
 $(\tilde{\phi}_{\tilde{x}})^2 = 1 + 2\varepsilon \phi_{1,\tilde{x}} + \varepsilon^2 \phi_{1,\tilde{x}}^2$
 $\tilde{\phi}_{\tilde{x}\tilde{x}} = \varepsilon \phi_{1,\tilde{x}\tilde{x}}$
 $\tilde{\phi}_{\tilde{y}} = \varepsilon \phi_{1,\tilde{y}}$
 $(\tilde{\phi}_{\tilde{y}})^2 = \varepsilon^2 \phi_{1,\tilde{y}}^2 \approx 0$

$\tilde{\phi}_{\tilde{y}\tilde{y}} = \varepsilon \phi_{1,\tilde{y}\tilde{y}}$
 $\tilde{\phi}_{\tilde{x}\tilde{y}} = \varepsilon \phi_{1,\tilde{x}\tilde{y}}$

$\Rightarrow$ substitute into governing eq. & BCs
 (drop all $\sim$ for brevity)

D1) Governing equations:

$$\left[M_\infty^2 + \frac{\kappa-1}{2} - \frac{\kappa+1}{2} (1 + 2\varepsilon \vartheta_{1,x}) \right] \varepsilon \vartheta_{1,xx} + \left[M_\infty^2 + \frac{\kappa-1}{2} \cdot (-2\varepsilon \vartheta_{1,x}) \right] \varepsilon \vartheta_{1,yy} - 2(1 + \varepsilon \vartheta_{1,x}) \varepsilon^2 \vartheta_{1,y} \vartheta_{1,xy} = 0$$

- cancel small terms $\varepsilon^2 \ll 1$, $\varepsilon^3 \ll 1$ and divide by ε :

$$\boxed{(1 - M_\infty^2) \vartheta_{1,xx} + \vartheta_{1,yy} = 0} \leftarrow \text{linearized eq. for small disturbances}$$

$\varepsilon \ll 1$ (9)

$\Rightarrow$ can be solved analytically $\rightarrow M_\infty > 1 \Rightarrow$ wave equation

$\downarrow M_\infty < 1 \Rightarrow$ Prandtl-Glauert transformation
 $\rightarrow$ Laplace eq.

$\Rightarrow$ singularity method

D2) Boundary conditions:

- far-field: $\tilde{\vartheta} = \tilde{X}$ as $\tilde{x}^2 + \tilde{y}^2 \rightarrow \infty$

$$\tilde{X} + \varepsilon \vartheta_1 = \tilde{X}$$

$$\underline{\vartheta_1 = 0}$$

(10)

- symmetry: $\tilde{\vartheta}_y = 0$ at $\tilde{y} = 0$ for $\tilde{x} \leq -1 \vee \tilde{x} \geq 1$

$$\underline{\varepsilon \vartheta_{1,y} = 0}$$

(11)

D2.1- asymptotic expansion of kinematic BC:

$$\phi_y(x, \pm \varepsilon h) = \pm \varepsilon h' \phi_x(x, \pm \varepsilon h) \quad (7)$$

- Taylor expansion: $\phi_y(x, \pm \varepsilon h) = \phi_y(x, 0^\pm) \pm \varepsilon h \phi_{yy}(x, 0^\pm) + \mathcal{O}(\varepsilon^2)$
 $\phi_x(x, \pm \varepsilon h) = \phi_x(x, 0^\pm) \pm \varepsilon h \phi_{xy}(x, 0^\pm) + \mathcal{O}(\varepsilon^2)$ (12)

- Substitute asymptotic expansion:

$$\begin{aligned} \phi_y(x, \pm \varepsilon h) &\approx \varepsilon \phi_{1,y}(x, 0^\pm) \pm \varepsilon h \varepsilon \phi_{1,yy}(x, 0^\pm) \xrightarrow{\varepsilon^2 \ll 1} \\ \phi_x(x, \pm \varepsilon h) &\approx 1 + \varepsilon \phi_{1,x}(x, 0^\pm) \pm \varepsilon h \varepsilon \phi_{1,xy}(x, 0^\pm) \end{aligned} \quad (13)$$

- Substitute ⁽¹³⁾ into the BC (7):

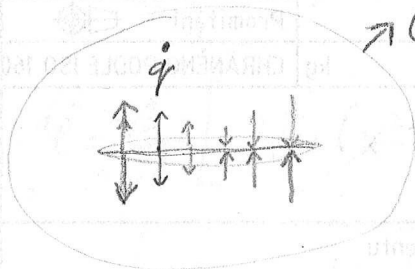
$$\varepsilon \phi_{1,y}(x, 0^\pm) = \pm \varepsilon h' \left[1 + \varepsilon \phi_{1,x}(x, 0^\pm) \right] \quad / : \varepsilon$$

$$\phi_{1,y}(x, 0^\pm) = \pm h' \quad (14)$$

discontinuous $\frac{\partial \phi_1}{\partial y} = 0$ at the airfoil

$\Rightarrow$ For the outer potential flow (excludes thin boundary layer) the airfoil is equivalent to a singular (infinitely thin) source/sink distribution at the axis.

$\nearrow \dot{Q} = 0$ zero net source $\Rightarrow$ mass is conserved



Flow described by (9, 10, 11, 14)

E) Analytical solution for $M_\infty < 1$: Prandtl-Glauert transform

$$\beta = \sqrt{1 - M_\infty^2}, \quad \bar{y} = \beta y, \quad \bar{\phi} = \beta^2 \phi_1 \quad (15)$$

- Governing eq: $\beta^2 \frac{\partial^2 \phi_1}{\partial x^2} + \frac{\partial^2 \phi_1}{\partial y^2} = 0 \rightarrow \frac{\partial^2 \bar{\phi}}{\partial x^2} + \frac{\partial^2 \bar{\phi}}{\partial \bar{y}^2} = 0$

$$\boxed{\Delta \bar{\phi} = 0} \quad (16)$$

- kinematic BC: $\frac{\partial \phi_1}{\partial y} = h' \rightarrow \boxed{\frac{\partial \bar{\phi}}{\partial \bar{y}} = \beta h'}$ (17)

at airfoil

$\Rightarrow$ equivalent incompressible flow ($M_\infty \rightarrow 0$) around $\bar{y}_p = \beta \varepsilon h(x)$

F) Solution via source distribution: singularity method

- For an incompressible potential flow due to a ^{line} source distribution, an analytical solution exists:

$$\bar{\phi} = \frac{1}{2\pi} \int_{-1}^1 \bar{q}(\xi) \ln \sqrt{(x-\xi)^2 + \bar{y}^2} d\xi \quad (18) \quad (\otimes)$$

- The source distribution is given by: $\bar{q}(x) = \bar{\phi}_5(x, 0^+) - \bar{\phi}_5(x, 0^-)$
 $= 2\bar{\phi}_5(x, 0^+) \quad (21)$

Example: consider a parabolic profile: $h = 1 - x^2$
 $h' = -2x \quad (22)$

$$\Rightarrow \bar{q}(x) = 2\beta h' = -4\beta x$$

$$(23)$$

$$\bar{\phi} = \frac{-2\beta}{\pi} \int_{-1}^1 \ln \sqrt{(1-\xi)^2 + \bar{y}^2} d\xi \quad (24)$$

→ ask Wolfram Mathematica for analytical sol.:

$$\bar{\phi} = \frac{\beta}{\pi} \left[2x - 2x\bar{y} \left(\arctan \frac{1-x}{\bar{y}} + \arctan \frac{1+x}{\bar{y}} \right) + (\bar{y}^2 + 1 - x^2) \operatorname{arctanh} \frac{2x}{1+x^2+\bar{y}^2} \right] \quad (25)$$

$$(*) \bar{u} = \frac{\partial \bar{\phi}}{\partial x} = \frac{1}{2\pi} \int_{-1}^1 \bar{q}(\xi) \cdot \frac{x-\xi}{(x-\xi)^2 + \bar{y}^2} d\xi \quad (19)$$

$$\bar{u} = \frac{2\beta}{\pi} \left[2 - \bar{y} \left(\arctan \frac{1-x}{\bar{y}} + \arctan \frac{1+x}{\bar{y}} \right) - x \operatorname{arctanh} \frac{2x}{1+x^2+\bar{y}^2} \right] \quad (26)$$

$$(*) \bar{v} = \frac{\partial \bar{\phi}}{\partial \bar{y}} = \frac{1}{2\pi} \int_{-1}^1 \bar{q}(\xi) \frac{\bar{y}}{(x-\xi)^2 + \bar{y}^2} d\xi \quad (20)$$

Substitution

$$\bar{v} = \frac{-2\beta}{\pi} \left[xa + \frac{1}{2} \bar{y} \ln \left(1 - \frac{4x}{(1+x)^2 + \bar{y}^2} \right) \right]; \quad \left[a = \arctan \frac{1-x}{\bar{y}} + \arctan \frac{1+x}{\bar{y}} \right] \quad (27)$$

G) Backward transformation:

$$\phi_1 = \frac{\bar{\phi}}{\beta^2} = \frac{1}{\beta\pi} \left[2x - 2\beta xy a + (\beta^2 y^2 + 1 - x^2) b \right] \quad (28)$$

$$\mu_1 = \frac{2}{\beta\pi} \left[2 - \beta y a - x b \right] = \frac{1}{\beta^2} \bar{\mu} \quad (29)$$

$$N_1 = \frac{-2}{\pi} \left[x a + \frac{1}{2} \beta y \ln \left(1 - \frac{4x}{(1+x)^2 + \beta^2 y^2} \right) \right] = \frac{1}{\beta} \bar{N} \quad (30)$$

H) Pressure : • Energy equation : $h + \frac{1}{2} \underline{u}^2 = \text{const.}$ (31)

• Thermodynamic relationships:
(adiabatic compression of ideal gas)

$$\frac{dT}{T} = \frac{\gamma-1}{\gamma} \frac{dP}{P} \quad (32) \quad (\text{p. 85 of Schneider, 2012})$$

for $\varepsilon \ll 1$: $T - T_\infty \ll T_\infty$
 $P - P_\infty \ll P_\infty \Rightarrow \frac{P - P_\infty}{P_\infty} = \frac{\gamma}{\gamma-1} \frac{T - T_\infty}{T_\infty}$ (33)

From energy eq. : $h - h_\infty = \frac{U^2 - \underline{u}^2}{2}$

$$c_p (T - T_\infty) = \frac{U^2 - \underline{u}^2}{2}$$

$$\frac{T - T_\infty}{T_\infty} = \frac{U^2 - \underline{u}^2}{2 h_\infty}$$

$$\frac{P - P_\infty}{P_\infty} = \frac{\gamma}{\gamma-1} \frac{U^2 - \underline{u}^2}{2 h_\infty}$$

from Brennen (2016), (Bon 12 - Bon 14)

$$h_\infty = \frac{\gamma}{\gamma-1} \frac{P_\infty}{\rho_\infty}$$

$$\frac{P - P_\infty}{P_\infty} = \frac{\rho_\infty}{P_\infty} \frac{U^2 - \underline{u}^2}{2} \cdot \frac{P_\infty}{\rho_\infty U^2}$$

← (Bernoulli equation)

$$\underbrace{\frac{P - P_\infty}{\frac{1}{2} \rho_\infty U^2}}_{\tilde{P}} = \frac{U^2 - \underline{u}^2}{2 U^2} \xrightarrow[\text{variables}]{\text{dimensionless}} \tilde{P} = \frac{1}{2} (1 - |\tilde{u}|^2)$$

H.1) Bernoulli equation for small disturbances:

$$|\underline{\tilde{u}}|^2 \approx (1 + \varepsilon |\underline{u}_1|)^2 = 1 + 2\varepsilon |\underline{u}_1| + \cancel{\varepsilon^2 |\underline{u}_1|^2}$$

$$\Rightarrow \boxed{\tilde{p} = -\varepsilon |\underline{u}_1|}$$

I) Streamlines:

- curves tangential to the flow velocity
- trajectories of fluid elements in a steady flow



$$\boxed{\frac{dy_s}{dx} = \frac{v}{u}}$$

← initial value problem (1st order ODE)

$$\boxed{y'_s(x) = f(x, y_s(x))}$$

- formal solution:

$$y_s(x) = y_0 + \int_{x_0}^x v(\bar{x}, y_s(\bar{x})) u^{-1}(\bar{x}, y_s(\bar{x})) d\bar{x}$$